#### **Chapter 7: Gas-Phase Concentration measurement**

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## Motivation for quantitative species concentration measurement



- Quantitative concentration measurements are motivated by various scientific and technological issues
- Target species depends on the combustion process
  - Mixture fraction
  - Reaction progress and intermediate species concentration
  - Pollutants
  - ...
- A quantitative species measurement in combustion application requires information of local gas temperature (density correction and methodrelated correction such as quenching correction in LIF)
  - Best option: measure simultaneously local temperature



# Methods in the NIR/UV/VIS for temperature measurements via Boltzmann distribution



- Laser absorption spectroscopy (LAS)
- Laser-induced fluorescence (LIF)
- Raman spectroscopy (RS)



#### Laser absorption spectroscopy (1)



• Experimental setup



 Deduce number densities from Beer-Lambert's law (shown in its simplest form)

$$I(L) = I_{0} \exp\left(-L \cdot N_{j} \cdot \sigma_{j \leftarrow i}\right) \Leftrightarrow N_{j} = \frac{\ln\left(\frac{I(L)}{I_{0}}\right)}{L \cdot \sigma_{j \leftarrow i}}$$

Calibration free (once the path length and absorption cross-section is known) Line-of-sight: no resolution along laser beam path

 $\rightarrow$  restrictions for application in turbulent flames

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## Tuneable diode laser absorption spectroscopy TDLAS



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#### Laser Induced fluorescence: Principle

Step 1: Absorption

- Electronic excitation of molecules by laser radiation
- Wavelength  $\lambda_A$
- Step 2: Spontaneous emission (fluorescence)
- Spectrally red-shifted  $\lambda_A < \lambda_E$
- Upper state lifetime few ns
- Measure of local number density
- Linear LIF regime

$$I_{LIF}(\mathbf{x}) = N_{1}(\mathbf{x})\sigma\gamma(\mathbf{v})I_{laser}(\mathbf{x})\frac{\tau_{tot}}{\tau_{sp}}U\frac{\Omega}{4\pi}\varepsilon\eta$$





# Laser Induced fluorescence: Experimental setup



• Experimental setup





Laser Induced fluorescence: pros and cons

- Good spatial resolution (90° detection angle)
- Sensitive
- Calibration required to determine  $U \frac{\Omega}{4\pi} \varepsilon \eta$
- Total lifetime  $\tau_{tot}$  needed but often not known!

$$\tau_{tot} = \frac{1}{A + P + Q}$$

- A: Einstein A-coefficient, molecular property, often known
- P: Predissociation rate, molecular property, often known
- *Q* : Quenching rate, depends on gas matrix, pressure, temperature  $\rightarrow$  PROBLEM

#### Total lifetime $\tau_{tot}$ makes quantitative LIF challenging!



$$I_{LIF}(\mathbf{x}) = N_1(\mathbf{x})\sigma\gamma(\mathbf{v})I_{laser}(\mathbf{x})\frac{\tau_{tot}}{\tau_{sp}}U\frac{\Omega}{4\pi}\varepsilon\eta$$

#### How to make LIF quantitative – options



- 1. Quantification of  $\tau_{tot}$ : Measure  $\tau_{tot} = fct(T, species)$ , once  $\tau_{tot}(T, species)$  is known measure LIF simultaneously with T and species (via Raman/Rayleigh)
- 2. Calibration: determine C = C(T) in  $I_{LIF}(x) = C(T)[N_1(x)]$  $\rightarrow$  Example CO-LIF in application example flame-wall interaction
- 3. Saturated LIF: does not really work, not detailed here
- 4. Combine 1D-LIF with absorption spectroscopy (see CST 158, 2000, Pixner et al.)

$$N_{1}(x) = -\frac{1}{\sigma} \cdot \frac{(T-1)\frac{dV(x)}{dx}}{(T-1)V(x)+1} \qquad V(x) = \frac{\int_{0}^{x} F(x')dx'}{\int_{0}^{L} F(x')dx'} = \frac{\exp\left(-\sigma\int_{0}^{x} N_{1}(x')dx'\right) - 1}{\exp\left(-\sigma\int_{0}^{L} N_{1}(x')dx'\right) - 1} \\ T = \frac{I(L)}{I(0)} = \exp\left(-\sigma\int_{0}^{L} N_{1}(x')dx'\right)$$



#### How to make LIF quantitative – options



5. Predissociative LIF: Independent of Q; low SNR, works for few molecules at low pressure



Inter nuclear distance



#### Raman spectroscopy



• Elastic and inelastic light scattering of photons off molecules





#### **Combined Raman/Rayleigh spectroscopy**



• Multi-scalar method: instantaneous measurement of main species (Raman) and temperature (Rayleigh)





#### **Combined Raman/Rayleigh spectroscopy**



• Multi-scalar method: instantaneous measurement of main species (Raman) and temperature (Rayleigh)





#### **Combined Raman/Rayleigh spectroscopy**



• Multi-scalar method: instantaneous measurement of main species (Raman) and temperature (Rayleigh)



- Spectral dispersion and simultaneous meas. of: CO<sub>2</sub>, O<sub>2</sub>, CO, N<sub>2</sub>, CH<sub>4</sub>, H<sub>2</sub>O, H<sub>2</sub>, equivalence ratio (phi), Temp.
- Challenges:
  - Low Raman scattering cross-sections and single-shot requirement
  - Data evaluation of noisy data
  - 1D application, high spatial resolution



# Raman spectroscopy: selection rules and spectra



• Selection rules

 $\Delta J = 0, \pm 2$   $\Delta J = 0 \rightarrow \text{Q-branch}$   $\Delta J = +2 \rightarrow \text{O-branch}$  $\Delta J = -2 \rightarrow S\text{-branch}$ 

#### Oxygen molecule $O_2$ , T = 1500 K

Simulated "stick spectrum" - infinite resolution

Ro-vibronic Stokes Raman

Exception: very weak R and P-lines



#### Raman spectroscopy: setup **Experimental setup** Fixed frequency laser filter lens objective Low dispersion spectrum +5.0e+4 N<sub>2</sub> experiment 1619 K Т - libary background CO2 0.036 libary + bgr O2 0.068 CO 0.042 intensity [a.u.] +5.26+4 0.656 $N_2$ CH<sub>4</sub> 0.025 H<sub>2</sub>O 0.129 H<sub>2</sub> 0.045 **Spectrometer** н,0 CH Η2 Shuttered CCD +0.0e+0 +5.0e+3 residuum +0.0e+0 -5.0e+3 560 600 640 680 wavelength [nm]



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#### 1D Raman/Rayleigh: spectrometer



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• New TU Darmstadt design





#### 1D Raman/Rayleigh: spectrometer









## 1D Raman/Rayleigh spectroscopy: iterative post-processing procedure



• Raman (inelastic) scattering  $\rightarrow$  concentrations  $N_i(r)$ 

$$S_{ram,i}\left(\vec{r}\right) \propto \sigma_{ram,i}\left(\frac{T(\vec{r})}{I_{Laser}} \quad N_{i}(\vec{r})\right)$$
Rayleigh (elastic) scattering  $\rightarrow$  temperature T(r)
$$S_{ray}\left(\vec{r}\right) \propto \sigma_{ray}\left(N_{i}(\vec{r})\right) I_{Laser} \sum N_{i}\left(\vec{r}\right)$$

$$\sigma_{ray} = \sum_{i} \left(\frac{N_{i}(\vec{r})}{\sum_{j} N_{j}(\vec{r})}\right) \sigma_{ray,i} \xrightarrow{\text{Ideal gas law}} T(\vec{r}) \propto \frac{1}{\sum N_{i}(\vec{r})}$$

 Determination of N<sub>i</sub>, T by iterative procedure: Need σ<sub>ram,i</sub> of each species *i*



#### Spectral fit 1619 K N<sub>2</sub> experiment т libary CO2 0.036 background 02 0.068 libary + bgr CO 0.042 N<sub>2</sub> 0.656 intensity [a.u.] CH4 0.025 H<sub>2</sub>O 0.129 H<sub>2</sub> 0.045 CH +0.0e+0 +5.0e+ residuum +0.0e+0 -5 Oe+: šer 600 640 680 Combining the strength of both wavelength [nm] Matrix inversion (MI) method $\rightarrow$ The Hybrid MI-method $\vec{S} = \underline{M} \square \vec{N}$ $\Rightarrow \overrightarrow{N} = \underline{\underline{M}}^{-1} \square \overrightarrow{S}$



#### **Different options for data evaluation**

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## A Hybrid Method for Data Evaluation in 1D Raman Spectroscopy

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- Spontaneous Raman scattering
  - important technique in combustion research
  - major species, scalar gradients, scalar dissipation
  - need best possible accuracy, precision, and spatial resolution
  - improvements in detection hardware and methods of analysis
- Two approaches to data analysis for hydrocarbon flames; both are complicated
- "Hybrid" method of Raman data analysis
  - combine strengths of methods used by Sandia and TU Darmstadt
  - reduce level of expertise needed to interpret Raman data
- Demonstrate using laminar flame measurements from Sandia system
  - Premixed CH<sub>4</sub>/air flat flames
  - Laminar H<sub>2</sub> jet flame (no hc fluorescence interference)





- On-chip binning of Raman bands to reduce camera readout noise
- Matrix equation relating signals and sources



- Extensive calibrations to determine temperature dependence of matrix elements for Raman response and crosstalk (represented as polynomials)
- Solve inverse problem to get species concentrations and temperature
- Iterate on Rayleigh temperature (1K conversion, 3-4 iterations)



## Spectral Fitting (TU Darmstadt)



- Individual rovibrational Raman transitions calculated for each species, based on Placzek's theory of polarizability (TUD Ramses code)
- Each Raman transition convolved with experimentally determined apparatus function, then all convolved Raman transitions superposed
  - Rayleigh scattering image  $\rightarrow$  apparatus function in this work
  - Raman bands broaden with increasing temperature due to the population of higher quantum states
  - Spectral library composed of temperature-dependent Raman bands
- The spectral library for each molecule is calibrated to an experimental spectrum measured in a gas sample with known mole fraction and temperature.
- Details of fitting procedure (Dirk Geyer thesis)







#### Matrix Inversion (old version)

- Pro's
  - Lower noise
  - Faster acquisition & processing
- Con's
  - Extensive calibrations required
  - Cannot calibrate accurately at some conditions
  - Spectral information lost
  - Impractical to account for beam steering or spatial dependence of response function

#### **Spectral Fitting**

- Pro's
  - Based on quantum mechanical theory
  - One calibration per species
  - Beam steering handle automatically
  - Background corrected more rigorous
- Con's
  - Higher readout noise
  - Slower data acquisition rate
  - Significant time/effort in fitting



### **Turbulent Combustion Laboratory:** Raman/Rayleigh/CO-LIF & Crossed OH PLIF





- T, N<sub>2</sub>, O<sub>2</sub>, CH<sub>4</sub>, CO<sub>2</sub>, H<sub>2</sub>O, H<sub>2</sub>, CO
- 6-mm segment
- Mixture fraction, reaction progress  $\geq$
- 3D flame orientation  $\geq$
- 1D, 3D scalar gradients, dissipation  $\succ$





### **Detection System**







• LSF 40-60 μm

**Transmission grating** 

- Mechanical gate: 3.9 μs gate (FWHM)
- 103 μm data spacing



### **Optical Bowing Effect**





Wavelength  $\rightarrow$ 

- Image of N<sub>2</sub> Raman scattering in air
- Low f-number spectrograph
  - bowed image of slit
  - Jun Zhao, Appl. Spectr. 57, 11 (2003)
     1368-1375
- Calculated (Zemax) and measured
  - Map CCD for wavelength at each pixel
- Must account for this optical effect



#### **Spectrometer Characteristics**





 Calculated (Zemax) dispersion is not linear across the image plane



 Spectral/spatial calibration using neon lamp and target with 50-µm holes on 200-µm centers



#### **Spectral Library vs. Measurement**





33rd International Symposium on Combustion, Beijing 2010







CRE

**Calculated Spectral Libraries (Ramses code)** 



#### **Major Species Response Curves**



CRE



- Good agreement, except for O<sub>2</sub> curve shape at high T
- O<sub>2</sub> response sensitive to bowing effect and beam steering

### Crosstalk between CO<sub>2</sub> and O<sub>2</sub>



CRE



- Greater uncertainty in calibration polynomials
- Sensitive to bowing effect and beam steering



- $O_2$  mole fraction should be zero in the rich flame products ( $\phi$ =1.3)
- Bowing effect leads to error of  $X_{O2} \sim 1.6\%$  due to uncorrected  $CO_2$  crosstalk

## Comparison for Laminar H<sub>2</sub> Jet Flame



- Close agreement on mean values from hybrid-MI and spectral fitting
- Hybrid method yields better precision (on chip binning)
- Spectral fitting yields lower noise than MI with software binning

CRE





- Hybrid method of Raman data analysis has been developed
  - Calculated Raman libraries are integrated to determine temperature-dependent terms for matrix inversion
  - Response and cross talk for  $N_2$ ,  $O_2$ ,  $H_2$ , CO, CO<sub>2</sub>,  $H_2O$
- Combines advantages from both previous methods
  - Low noise from on-chip binning
  - Fast data acquisition and processing
  - Temperature dependence based on QM theory
  - Automatic correction for image bowing and beam steering
- Relatively easy to adapt to other Raman/Rayleigh systems





#### **Application example**



 Thermo-chemical state in swirling premixed flame: Gregor et int. Dreizler. Proc. Combust. Inst. 32, 1739 – 1746 (2009)



## Thermo-chemical state in swirling premixed flame



• Operational conditions

		PSF-30	PSF-90	PSF-150
S <sub>0,th</sub>	[-]	0.75	0.75	0.75
Ρ	[kW]	30	90	150
$\phi$	[-]	0.833	0.833	1.0
Re <sub>tot.</sub>	[-]	10000	29900	42300



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#### **Measurement locations**



Radial profiles at 4 axial heights







#### Raman scattering – exp. setup



... from delay line • *Raman:* gating by a rotating shutter (~  $10 \mu s$ ) suppressing spurious light Czerny-Turner type imaging ID probe volume spectrometer beam waist ~ 400 μm thermo-electrically cooled backside illuminated CCD energy referencing steep cutoff long pass filter reflecting Rayleigh scattered light / transmitting Raman scattered light

#### **TECHNISCHE** Raman scattering – exp. setup UNIVERSITÄT DARMSTADT Data evaluation is conducted with a spectral fitting method using theoretical Raman 0 spectra Thermo-kinetic state Y, transient T = 1606 K measured 500 Lib & background CO<sub>2</sub> 0.039 weighted O<sub>2</sub> 0.030 400 N<sub>2</sub> residuum CO 0.064 N2 0.701 Intensity [-] 300 CH\_ 0.024 H<sub>2</sub>O 0.104 H<sub>2</sub>O 0.039 CH. H, 200 CO\_02 CO 100 Н 0 energy 560 600 640 680 Wavelength [nm]



# Raman scattering – spatial resolution

 using highly resolved (77.5 µm) Rayleigh signal to determine radial instantaneous temperature profiles

310 µm

max. 400

μm

 using coarsely resolved Raman signal (310 µm) to measure species concentrations & correct for scatter cross section Raman probe volume

77.5 µm

Rayleigh

probe

volume







#### **Radial Profiles (mean and rms)**





- inner recirculation zone (~0-10mm):
  - fully burnt exhaust
  - heat loss to bluff body





#### **Radial Profiles (mean and rms)**



z = 10 mm2000 mean 1500 rms ∑<sup>1000</sup> 500 0 1,00 mean eduix. 0,75 eduix. 0,50 eduix. rms bluff body i flange 0.00 20 30 10 50 40 0 r [mm]

- inner recirculation zone (~0-10mm):
  - fully burnt exhaust
  - heat loss to bluff body
- inner mixing layer (~10-20mm):
  - strong intermittency hence high fluctuation levels
  - r ~ 17 mm: mean premixed flame front position as deduced from max. T & c gradients / max. CO concentration



#### **Radial Profiles (mean and rms)**



z = 10 mm2000 mean 1500 rms ∑<sup>1000</sup> ⊢ 500 0 1,00 mean edniv. ratio phi [-] 0,75 edniv. ratio phi 0,50 rms bluff body flange 0.00 30 10 20 40 50 0

r [mm]

- inner recirculation zone (~0-10mm):
  - fully burnt exhaust
  - heat loss to bluff body
- inner mixing layer (~10-20mm):
  - strong intermittency hence high fluctuation levels
  - r ~ 17 mm: mean premixed flame front position as deduced from max. T & c gradients / max. CO concentration
- central annular jet (~20-27mm):
  - almost unreacted fuel without dilution
- outer mixing layer (~27-38mm):
  - steep phi gradient towards surrounding air (phi = 0)
  - second temperature maximum due to outer recirculation zone



#### Mean Radial Profiles - z Dependency





- for increasing height z:
- mean flame front position at larger radii
- wider flame brush & broader mixing layer hence flatter radial T & phi profiles
- distinct 2nd T maximum (up to z = 30mm) due to outer recirculation of exhaust
- premixed flame front is located in regions with very low stratification of nearly constant phi (up to z = 30mm)





#### **Single Shot Correlations - Scatter Plots**

- single shot correlation between 2 or more scalars (e.g. T & CH4)
- each scatter corresponds to one single shot
- samples spanned between 3 thermo-kinetic states:
  - a) hot burnt exhaust
  - b) cold unburnt fuel
  - c) cold secondary air



#### TECHNISCHE **Single Shot Correlations - Scatter Plots** UNIVERSITÄT DARMSTADT <del>0,10</del> (0-10mm) IRZ samples can be assigned to flame 0,08 (10-20mm) IML regions: CAJ (20-27mm) 0,06 inner recirculation zone (0-10mm): OML (27-38mm) $\mathsf{X}_{\mathsf{CH4}}$ fully burnt samples @ max T, no 0,04 CF (>38mm) secondary air 0,02 inner mixing layer & central annular jet (10-27mm): 0.00 burnt, unburnt or mixed samples, no secondary air 1,2 outer mixing layer (27 - 38 mm): 1,0 mostly unburnt fuel and / or 0,8 secondary air phi 0,6 rarely mixing between burnt 0,4 samples and secondary air 0,2 coflow / flange (>38 mm):

0,0

400

800

1200

T [K]

exclusively ambient air



2000

1600

#### **Scatter Plots - z Dependency**



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- at z = 30 mm:
- significantly increased probability to measure intermediate samples due to:
  - a) secondary air entrainment > into burnt exhaust (OML, CF)
  - b) mixing between burnt / unburnt fuel (CAJ) and reaction without dilution
  - c) mixing of air with reacted AND unreacted fuel

or

 slowed or extinguished reactions (CAJ, OML, CF)





# Conclusions Raman/Rayleigh scattering in swirling premixed flame



- Temperature is key-quantity (reaction progress)
- Thermo-kinetic state precisely measured
- Thermo-kinetic state is a prerequisite for understanding pollutant formation and finite rate chemistry effects
- Main findings are
  - In region of flame stabilization (z < 30 mm) premixed flame front is not located in areas of stratification but in areas of almost constant phi
  - typical reaction and / or mixing behavior can be assigned to different flame regions
  - intermediate reaction states are promoted by dilution with air
  - distinction between pure mixing and local flame quenching needs additional diagnostics monitoring intermediate species

